

acquired for approaches to Gainesville Regional Airport (GNV) using commercial airborne lidar systems with standard topographic mapping configurations and compared against field-surveyed reference data. Although several field-surveyed obstructions, such as poles and antennas, were missed in the lidar surveys, the study yielded the following important conclusions (Anderson et al. 2002; Parrish 2003):

- (1) The limiting factor in the ability to meet FAA specifications for obstruction data with lidar is not the accuracy with which laser returns can be geolocated. Rather, the primary factors are the ability to achieve sufficient horizontal and vertical point density and sufficient received signal strength to detect the tops of small-diameter, low-reflectance objects, such as poles and antennas.
- (2) The typical system configurations and data acquisition and processing parameters used for topographic mapping (i.e., generation of bare-earth terrain data sets) are poorly suited for airport obstruction surveying.

These conclusions motivated a second phase of research, in which the capability to improve obstruction detection results through appropriate selection of the system settings and data acquisition and processing parameters was investigated (Parrish et al. 2005). The data acquisition included flights using fourteen different configurations, consisting of different settings and combinations of forward tilt angle, beam divergence, and flying height. The two best configurations resulted in 100% detection of the field-surveyed objects at GNV. A third phase of NGS' research into lidar airport obstruction surveying was then carried out in 2003 to investigate the capability to conduct a complete end-to-end obstruction survey using lidar (Parrish et al. 2004). Data were acquired using a dual lidar system (one sensor head at nadir and the other configured with a 20° forward tilt angle) for Frederick Municipal Airport (FDK) and

Stafford Regional Airport (RMN). A noteworthy outcome of this work was the development by the FAA of the first instrument approach procedure in the U.S. based on a lidar survey (Fred Anderson, internal FAA report, 2004).

While the above-mentioned studies made great strides towards demonstrating the technical feasibility of lidar obstruction surveys, the issue of practicality remained largely unaddressed. A major challenge in performing lidar obstruction surveys efficiently and cost effectively stems from the fact that, although the FAA is migrating towards acceptance of survey data in a variety of Geographic Information System (GIS) compatible formats (FAA 2008), the current operational workflow and software used by FAA approach procedure developers are not designed to work with raw lidar point clouds. Rather, they require obstruction data in the form of discrete obstacle lists containing coordinate information and feature names for individual objects (e.g., trees, poles, buildings, etc.) in text-based Universal Data Delivery Format (UDDF) files (FAA 2007; FAA 1996). Creating these final UDDF obstruction lists from the raw point cloud is a non-trivial task. Using NGS surface modeling software, it is possible to analyze the raw lidar point cloud against the OIS and extract penetrating lidar points (Parrish et al. 2004). However, the removal of non-penetrating lidar points complicates all of the following tasks:

- (1) distinguishing between returns from actual objects and those due to noise or clutter;
- (2) determining which points correspond to returns from the same object; and
- (3) attributing detected obstructions.

Hence, previous NGS lidar airport obstruction surveys required a great deal of manual analysis, largely negating the intended cost and time savings. An independent study by Wang et al. (2004) described a workflow for lidar airport surveying based on National Geospatial-Intelligence Agency (NGA, formerly the National Imagery and Mapping Agency) requirements. As in the

NGS studies, the end process was fairly complicated, containing approximately 20 steps and relying on aerial imagery and topographic maps.

In this paper we describe a new workflow for lidar airport obstruction surveying that addresses the problems described above. Two important aspects of our approach are the utilization of the additional information contained in full-waveform lidar data to enhance vertical object (VO) detection and the sequencing of steps in the workflow to improve both the efficiency and end results. Experiments conducted using lidar waveform data collected for two project sites in Madison, Wisconsin, are described, followed by the results and conclusions.

Proposed Workflow for Lidar Airport Obstruction Surveying

The lidar airport obstruction surveying workflow developed and tested in this research is illustrated in Fig. 1. This workflow has been specifically designed and ordered to form an efficient practical methodology for lidar obstruction surveying, while simultaneously improving the capability to meet FAA obstruction surveying standards. As shown in the figure, it comprises four steps, with the input consisting of full-waveform lidar data (which will be further described below) and aerial imagery, preferably, but not necessarily, collected concurrently. The output consists of final obstruction lists in the FAA-specified format. Advantages of this workflow can be summarized as follows:

- Utilizes the additional information contained in full-waveform lidar to improve object detection.
- Minimizes the probability of a missed obstruction, because the detection threshold in the object extraction step can be set very conservatively, and false alarms can be easily (usually automatically) eliminated in the subsequent processing steps.

- Facilitates creation of the final FAA-required obstruction lists by placing the object extraction step as early in the process as possible and ensuring that all following steps operate on objects, not points.
- Increases efficiency by combining all steps that require manual analysis.
- Requires only four steps for generation of the final deliverables to the FAA (i.e., obstruction information in UDDF format).

Two assumptions are made regarding the input data: first, the airborne data acquisition has been performed using the recommended mission parameters and system settings contained in NGS (2004) and Parrish et al. (2005); second, the lidar data include digitized waveforms. Waveform digitization is a relatively new advancement in commercially-available airborne lidar systems (e.g., Hug et al. 2004; Gutierrez et al. 2005). Unlike earlier discrete-return systems that only recorded a small number of returns (e.g., one to four) per transmitted pulse, full-waveform systems are capable of digitizing the entire received signal from each transmitted pulse at very high sampling frequencies (e.g., 1 GHz), often producing a few hundred samples per return. Full-waveform systems overcome the “minimum pulse separation limitation” of discrete-return systems (i.e., the inability to resolve features separated by less than a few meters in the range direction), which results from the inevitable limitations of the rangefinder electronics (Nayegandhi et al. 2006; Optech 2006).

To clarify the above statements and more fully describe the workflow, the following is a detailed, step-by-step description of the process:

Step 1: Deconvolve and Georeference the Lidar Waveform Data

Fig. 2 shows examples of digitized waveforms collected using an Optech, Inc. Airborne Laser Terrain Mapper (ALTM) 3100 and waveform digitizer with a 1 ns sampling period. On the left hand side of the figure, photos of two VOs are shown: a tall TV broadcast tower/antenna (top) and a power transmission tower (bottom). The middle images show the locations of the objects on an orthophoto (denoted by the small “+” symbol), and on the right are corresponding lidar waveforms (i.e., digitized return signals from laser pulses incident on some portion(s) of these objects). The waveform plots are divided into three sections separated by lines parallel to the ordinate axis: the first (leftmost) section contains 40 samples of the transmitted (T_0) pulse, and in these examples is saturated or clipped; the second (middle) section contains the first 220 digitized samples of the return; and the third (rightmost) section contains the last 220 digitized samples of the return, if available. The digitized return waveform from the TV broadcast tower/antenna exhibits two peaks, separated by approximately 2097 ns (there is a gap of 1870 ns between the two sections of the return) or a range difference of approximately 314 m, accounting for the round trip travel time of the laser pulse. Based on visual analysis and this large difference in range, the two peaks in the return waveform appear to correspond to reflections from somewhere near the top of the tower and the ground. Similarly, the return waveform in the bottom portion of the figure contains three peaks, likely corresponding to reflections from two of the crossarms of the transmission tower and from the ground. (The return waveform shown in the bottom graph does not include a second section.)

Even from cursory visual analysis, it is apparent that these return waveforms contain valuable information; the question, however, is how to exploit the information for our intended application. For improved airport obstruction detection, we desire a very dense, detailed X,Y,Z,I

lidar point cloud (where X , Y , and Z are spatial coordinates, and I denotes intensity), in which the vertical structures of objects are well characterized. Each individual laser reflection should map to a point in the point cloud and the intensity value of the point should be proportional to the amount of backscattered energy. The process of generating this point cloud would be trivial if we were to start with a set of “ideal” waveforms, as depicted on the left-hand side of Fig. 3. This ideal waveform consists of a train of spikes in time, where each spike corresponds to an individual laser reflection, with its amplitude being proportional to the strength of the return and its position on the time axis being proportional to the range. For example, in a forested area, the first spike in a particular idealized return might correspond to a reflection from the top-most branch or leaf in a tree, while the second spike might correspond to a reflection from the next lower branch, and so on down to the ground.

An actual received waveform, as depicted on the right-hand side of Fig. 3, can be considered a blurry, noisy version of the ideal waveform. One method of estimating the desired parameters (e.g., individual ranges and amplitudes) from the observed waveform is through a process called Gaussian decomposition, originally proposed as a processing strategy for data from the large-footprint, NASA-owned-and-operated Laser Vegetation Imaging Sensor (LVIS) system (Hofton et al. 2000) and recently applied to data from some of the very early commercial, small-footprint, full-waveform systems (e.g., Persson et al. 2005; Reitberger et al. 2006; Wagner et al. 2006). In this approach, each received waveform is modeled as a convex combination of K Gaussian density functions (i.e., a K -component Gaussian mixture model), or

$$\hat{y}(t) = \sum_{i=1}^K \alpha_i \exp\left\{-\frac{1}{2\sigma_i^2}(t - \mu_i)^2\right\} \quad (1)$$

where α_i is the amplitude of the i^{th} component, σ_i is its standard deviation, and μ_i is its mean. The objective is to obtain statistically optimal estimates (based on some defined optimality

criterion) of the unknown parameters $\{\alpha_i, \sigma_i, \mu_i\}$ for K components, where K can be adjusted until the fit of the model to the data is within some prescribed tolerance. A preprocessing step of denoising by lowpass filtering is typically employed.

While some studies have demonstrated quite good results using Gaussian decomposition (e.g., Reitberger et al. 2006; Persson et al. 2005), in this research, we investigated a deconvolution approach. Although deconvolution of airborne lidar waveforms has received comparatively little attention in the literature to date (Nordin 2006; Walter 2005), the deconvolution strategy investigated in this work offers the following advantages over the Gaussian decomposition approach:

- (1) The algorithm is robust against local minima problems and, with slight modification to the thresholding step that will be described below (i.e., through use of a soft threshold) guarantees convergence to the globally-optimal solution.
- (2) The issue of noise in the data is addressed directly, rather than with an ad hoc preprocessing step.
- (3) The inverse problem is solved, rather than attempting to directly fit the return waveform data.
- (4) The computational costs are low.

Specifically, we employed the Expectation Maximization (EM) deconvolution strategy of Figueiredo and Nowak (2003), also referred to in the literature as an iterative shrinkage/thresholding (IST) algorithm. We start with the following observation model

$$y[n] = g[n] * x[n] + w[n] \quad (2)$$

where $x[n]$ is the ideal/undegraded (discrete-time) signal illustrated graphically on the left hand side of Fig. 3, $y[n]$ is the observed waveform depicted on the right hand side of the figure, $g[n]$ is

an approximately Gaussian impulse response function, $w[n]$ is additive white Gaussian noise (WGN) with variance σ_w^2 , and $*$ denotes convolution. The impulse response function, $g[n]$, depends on the laser pulse characteristics, system optics, atmospheric conditions at the time of data acquisition, and other variables, and can be easily determined in practice by simply measuring the received signal from a (near) perfect reflecting surface, such as flat water or concrete. The received signal from such a surface will consist of one “bump,” corresponding to the single reflection off the surface. Averaging several returns of this nature provides an approximately noise-free estimate of the impulse response.

The problem of determining the ideal waveform is challenging, since the deconvolution process is typically very ill-conditioned. To address this challenge, the deconvolution problem can be formulated as a Bayesian estimation problem. Selecting a “hit-or-miss” cost function yields the maximum a posteriori (MAP) estimator

$$\hat{\mathbf{x}} = \arg \max_{\mathbf{x}} p(\mathbf{x} | \mathbf{y}) = \arg \max_{\mathbf{x}} p(\mathbf{y} | \mathbf{x})p(\mathbf{x}) = \arg \max_{\mathbf{x}} \{ \ln p(\mathbf{y} | \mathbf{x}) + \ln p(\mathbf{x}) \} \quad (3)$$

where \mathbf{x} and \mathbf{y} denote vectors formed from the samples of $x[n]$ and $y[n]$, respectively, $p(\mathbf{x} | \mathbf{y})$ is the posterior probability density function (pdf) (i.e., the pdf of the unknown signal, \mathbf{x} , after the data, \mathbf{y} , have been observed) and $p(\mathbf{x})$ is the prior pdf (i.e., the pdf of \mathbf{x} before any data are observed). The aim of the prior pdf is to incorporate a priori knowledge of the salient characteristics of \mathbf{x} in order to stabilize the inversion process.

If we begin by assuming \mathbf{x} and \mathbf{y} to be jointly Gaussian distributed, then it follows that (Kay 1993):

- (1) The marginal pdf, $p(\mathbf{x})$ is Gaussian.
- (2) The posterior pdf, $p(\mathbf{x} | \mathbf{y})$, is also Gaussian.

(3) The MAP estimator and minimum mean square error (MMSE) estimator are the same, since the mean and mode (location of the maximum) of a Gaussian are identical.

In this case, the MAP estimator is equivalent to the well-known Wiener filter (linear MMSE estimator). (As an aside, Wiener filtering as a strategy for lidar waveform processing was discussed in Jutzi and Stilla (2006), although in that work it was utilized primarily as a denoising step preceding Gaussian decomposition of the signal.)

While the Wiener filter is a commonly-used approach for signal/image restoration, it suffers some serious drawbacks for this particular application. Specifically, the prior pdf assumed in the Wiener filter essentially models the signal, \mathbf{x} , as a white Gaussian noise process with a certain power, σ_x^2 , and does not reflect prior knowledge about the structure of the waveforms. However, based on the physics of the problem, we know our signals are not arbitrary, noise-like waveforms; rather they are much better modeled as trains of spikes in time, with each spike corresponding to an individual laser reflection (c.f. Fig. 3). The advantage of the approach developed by Figueiredo and Nowak (2003) is that it solves for the MAP estimate (Eq. (3)) while allowing us to incorporate knowledge of the waveform structure through selection of the prior pdf. The prior pdf employed here models the ideal waveforms as random spike trains, and is essentially determined by specifying the probability, p , of observing a spike at each point in time. In this model, each time point is treated as statistically independent of all other time points (i.e., the probability of a spike at one time point is independent of all other time points). This prior pdf is quite different from the Gaussian white noise pdf implicit in the Wiener filter. Because a closed form solution for the MAP estimator does not exist, it is computed numerically, iterating between a partial deconvolution step (the E-step) and a thresholding step (M-Step), which, in our implementation, can be expressed in the time domain as

$$\begin{aligned}
\text{E - step: } \hat{z}^{(t)}[n] &= \hat{x}^{(t)}[n] + g^*[n] * (y[n] - g[n] * \hat{x}^{(t)}[n]) \\
\text{M - step: } \hat{x}^{(t+1)}[n] &= \frac{\max\left\{\left(\hat{z}^{(t)}[n]\right)^2 - \tau\sigma_w^2, 0\right\}}{\hat{z}^{(t)}[n]}
\end{aligned} \tag{4}$$

In Eq. (4), $\hat{x}^{(t)}[n]$ is the estimate of the signal $x[n]$ at the t^{th} iteration of the algorithm; $\hat{z}^{(t)}[n]$ is the estimate of the missing data in the EM procedure (as explained in Figueiredo and Nowak 2001, 2003; Nowak and Figueiredo 2001); $g[n]$ is the impulse response function, estimated as described above; and, at the implementation level, the nonnegative parameter τ can be considered a tunable parameter controlling the number of spikes in the output. In particular, τ can be inversely related to the prior probability of a spike at a given location. The specific form of the denoising M-step in Eq. (4) results from the selection of the prior described in Figueiredo and Nowak (2003). The procedure above is iterated until convergence, which can be declared using any number of reasonable stopping criteria, such as stopping when the relative difference between successive iterations is less than a specified tolerance of 10^{-3} . It is important to note that each iteration can be rapidly computed. The E-Step requires two convolutions with the impulse response function. This can be computed in $N \log N$ operations (where N is the number of samples in the received signal) using the Fast Fourier Transform (FFT). The M-Step applies a scalar thresholding operation to each element of the output of the E-Step, and hence requires only N operations. Thus, the overall computational complexity of each iteration is $O(N \log N)$, roughly the same complexity as the Wiener filter.

Fig. 4 shows examples of deconvolving two actual lidar waveforms (top, green) using the EM deconvolution algorithm (bottom, red) and, for comparison, a Wiener filter (middle, blue). From these examples, the superiority of the EM deconvolution algorithm in this application is clear: the output signals are much sharper (in fact, we have produced the desired train of spikes in time), and the noise has been completely suppressed. Additionally, the EM deconvolution

algorithm output does not contain the ringing (artificial oscillations around sharp edges) evident in the Wiener-filtered waveforms—especially important, since, if the Wiener-filtered waveforms are subsequently georeferenced, the ringing artifacts carry over into the output point clouds (Parrish 2007a).

Since the location on the time axis of each spike in the graphs on the bottom portion of Fig. 4 is proportional to the range, and since the post-processed Global Positioning System/Inertial Measurement Unit (GPS/IMU) and scanner angle data are presumably already available through the standard processing, point clouds can be generated at this point through straightforward application of the laser geolocation equation (e.g., Vaughn et al. 1996; Filin 2001). The amplitude of each spike is the value assigned to the intensity of the corresponding point in the output point cloud.

An extremely attractive feature of this EM deconvolution algorithm is provision of the tunable parameter τ , capable of being adjusted based on the considerations of the particular end-user application. Decreasing τ has the effect of increasing the number of spikes in the output (and, hence, the density of the output point clouds), but also increases the probability of “false spikes”; increasing τ has the opposite effect. For our application of airport obstruction surveying, it is highly desirable to decrease τ to produce very dense output point clouds to assist in minimizing the probability of a missed obstruction. The increased probability of “false objects” output from the next step in the workflow turns out not to be a major concern, since these false objects can be easily (usually automatically) eliminated in the final two steps, as will be discussed.

Step 2: Detect Vertical Objects (VOs)

The vertical object detection (Step 2) can be performed using any of a number of algorithms or approaches, including a Hough transform to detect vertical cylinders, cones, and spheres; morphological filtering algorithms (Woods et al. 2004); a 3D wavelet domain approach (Parrish 2007a,b); or approaches based on eigenvalues of covariance matrices for spherical and cylindrical environments (Gross et al. 2007). While analysis and comparison of different object detection strategies is beyond the scope of this paper, two aspects of the detection are of critical importance in the application of lidar obstruction surveying, regardless of the specific algorithm utilized: 1) the data model used to store and operate on the lidar data, and 2) the selection of the detection threshold. Regarding the data model, the primary concern is avoidance of surface representations, such as 2D elevation images/grids, commonly referred to as digital elevation models (DEMs) or digital surface models (DSMs), or triangulated irregular networks (TINs). These representations all suffer a serious drawback for our application: namely, it is impossible to adequately model true vertical surfaces, convex or concave vertical surfaces, and objects containing multiple, discrete components at the same (X,Y) location but different heights, all of which leads to severe loss of information about vertical structure (Stoker 2004). For example, it is impossible to accurately represent roof overhangs, building sidewalls, transmission towers with multiple crossarms, water towers, and complex-shaped tree canopies. Furthermore, these surface representations greatly complicate the task of distinguishing between a return from an actual vertical object, such as a pole, and a return due to noise or clutter (e.g., a bird); in the elevation image representation both appear as bright pixels (higher elevation) surrounded by darker pixels (lower elevation). Two suitable alternatives to the use of surface representations

are: 1) raw point clouds, or 2) volume, or “voxel,” representations (e.g., Stoker 2004; Vosselman et al. 2004); the VO detection algorithm should operate on lidar data in one of these formats.

The second consideration is the detection threshold. In signal detection, there are always tradeoffs between probability of detection, P_D , and probability of false alarm, P_{FA} . A common method of displaying and analyzing the pattern of tradeoffs for a particular detection algorithm is through the so-called receiver operating characteristic (ROC) curve, a plot of P_D versus P_{FA} as the detector threshold is varied. Although ROC curves vary from detector to detector (and, therefore, are frequently used in comparing different detection algorithms), there exists a general trend: decreasing the detector threshold increases both P_D and P_{FA} . The sequence of steps in our workflow has been ordered such that, irrespective of the specific detection algorithm used, the detection threshold can be set very low to minimize the probability of a miss (or, equivalently, maximize P_D). Object detection is performed first, followed by the OIS analysis, followed by quality assurance and quality control (QA/QC) and object-type attribution. False alarms (or “false objects”) due to ground clutter or noise and randomly distributed throughout the project area will typically be automatically eliminated in the OIS analysis step, due to not penetrating or falling outside the OIS, and the remainder can be easily removed during the attribution and QA/QC step. This inherent tolerance of conservative detection thresholds in the object extraction step is an important aspect of our workflow, since the key consideration in airport obstruction surveying is to avoid missed obstructions that could potentially jeopardize flight safety.

Step 3: Analyze Detected Objects against OIS

The OIS analysis can be performed easily and quickly using NGS' Surface Model Library (SML), a downloadable software library that enables users to mathematically model an airport's OIS and perform related analysis (NGS 2008). SML is available as a dynamic-link library (DLL) that is compatible with Microsoft Visual Basic (VB), C++, and .NET. The library comprises dozens of functions, which allow users to calculate surface penetrations, analyze features relative to specified OIS, and numerous other related tasks. NGS is responsible for updating SML as needed to reflect any changes in the definitions of the OIS (FAA 2008).

Since the implementation is straightforward, the only aspect of this step that merits additional discussion is its order in the overall workflow; it is critical that the OIS analysis be performed after, not prior to, the object detection. The reason for this is threefold. First, as already noted, performing OIS analysis on raw point clouds to extract penetrating points greatly complicates tasks such as distinguishing between returns from actual objects and noise, grouping returns from the same object, and performing attribution. Second, this sequence enables the detection threshold in Step 2 to be set very conservatively (see Step 2 discussion above). Third, since the final deliverable to the FAA consists of information for obstructions (i.e., objects, not individual laser reflection locations), it is beneficial to extract objects as early in the process as possible and to ensure subsequent processing steps operate on those objects.

Step 4: Perform Object-Type Attribution and QA/QC in a Softcopy Photogrammetry or GIS Environment

While some previous studies have shown that classification (e.g., of tree species) can be improved using full-waveform data (Reitberger et al. 2006), in airport obstruction surveys, fully-

automated classification is arguably an unrealizable objective, because airport obstructions can include literally any type of natural or manmade object. Fortunately, in this application, manual attribution can be done without significantly increasing time or cost of the survey, since manual analysis is already required in the QA/QC step to meet the stringent quality assurance controls. As shown in previous phases of NGS' research into lidar obstruction surveying (e.g., Parrish et al. 2004), this QA/QC is most easily and efficiently performed using stereo imagery in a softcopy photogrammetry environment, or somewhat less desirably, using orthoimages viewed monoscopically in a GIS environment. (An advantage of the stereo option is that it enables an independent check of an object's top elevation.) By combining of all steps requiring manual analysis (i.e., attribution and QA/QC), as shown in Step 4 of Fig. 1, an efficient workflow that minimizes human time is enabled.

Experiment

The workflow described above was tested using airborne and field data for two project sites in Madison, Wisconsin, as shown in Fig. 5. The airborne data consisted of lidar waveform data collected with an Optech, Inc. Airborne Laser Terrain Mapper (ALTM) 3100 and waveform digitizer on June 23-24, 2006, as well as high-resolution digital aerial imagery acquired with an Applanix, Inc. Digital Sensor System (DSS) 322 on September 14-15, 2006. Received signals were split in the lidar system, enabling simultaneous acquisition and recording of both discrete-return and full-waveform data. Mission parameters for the airborne lidar and digital imagery data acquisition are shown in Tables 1 and 2, respectively. The field data consisted of 35 field-surveyed reference objects in each of the two project sites (70 objects total). The motivating factors in selecting these sites included:

- (1) numerous objects representative of “typical” airport obstructions, including buildings, towers, antennas, poles, and trees of differing heights and species;
- (2) ease of access by field personnel; and
- (3) existing ground control.

Since neither site contained an actual airport, for purposes of this study, we simulated a 1,829 m (6,000 ft) runway with a 6,096 m (20,000 ft) Area Navigation Approach (ANA) surface for a Localizer Performance with Vertical Guidance (LPV) approach in the southwest project site, as shown in Fig. 5.

Several preprocessing steps were necessary before initiating the workflow depicted in Fig. 1. First, the airborne kinematic GPS observables and IMU data were post-processed to create a “blended navigation solution” using POSPac v4.2 (Applanix, Inc.). The discrete-return (non-waveform) lidar data were processed using REALM v3.5.3 (Optech, Inc.). Additionally, 0.5-m resolution orthophotos were generated from the DSS imagery using RapidOrtho v4.3 (Applanix, Inc.) and a U.S. Geological Survey (USGS) 1/3-arcsecond DEM to correct for relief displacement. It would certainly have been possible to perform the orthorectification using a DEM derived from the lidar data. However, the USGS DEM was determined to enable sufficiently accurate orthophotos for use in Step 4, thus obviating the need for an additional DEM generation step.

Following the initial preprocessing, the workflow depicted in Fig. 1 was executed. The waveform deconvolution (Step 1) was carried out using the EM deconvolution algorithm described above and expressed mathematically in Eq. (4). The VO detection (Step 2) was performed using the algorithm described in Parrish (2007a). The OIS analysis (Step 3) was completed using NGS’ SML software. The input to the manual QA/QC and object-type

attribution step (Step 4) consisted of both the orthophotos generated from the DSS imagery and the “raw” stereo imagery, imported into ArcGIS v9.1 (ESRI, Inc.) and a Leica, Inc. stereo viewer, respectively. Final output consisted of obstruction lists in UDDF format, suitable for comparison against the field-surveyed reference data, or in an actual obstruction survey, delivery to the FAA.

Analysis

The objective of the analysis was to investigate and quantify the ability to improve and streamline generation of final FAA-formatted obstruction data sets using the workflow depicted in Fig.1. The three specific topics of investigation included:

- (1) the achievable densification of point clouds on and around airport obstructions using the full-waveform data;
- (2) the effectiveness of the workflow in automatically eliminating any “false objects” (thereby enabling conservative VO detection, while maintaining efficiency); and
- (3) the percentage change in end-to-end processing time using this workflow, as compared with previous NGS airport obstruction surveys.

To investigate the achievable densification of point clouds for obstructions using the full-waveform data, the point clouds output from Step 1 were compared against the point clouds output from REALM using the discrete-return data. The first step in the analysis was to create point cloud subsets around each of the field-surveyed objects in the southwest project area and to use the “Measure Point Density” tool in TerraScan v5.005 (TerraSolid, Ltd.) to count the number of points on each object. This semi-automated analysis was supplemented by manual analysis, which entailed visually examining the output point clouds for each object in profile mode in

TerraScan to compare both the relative point densities and the relative amounts of noise. The results of this analysis were quite promising: the use of the full-waveform data led to a 252% increase in the average number of points on the objects, as shown in Table 3, with very slight increase in the amount of noise in the point clouds on or near the objects, as seen in Fig. 6. This figure shows photos (left column) and profile views of point cloud subsets for two different VOs, a tree and TV broadcast tower, generated using the discrete-return lidar data (middle column) and full-waveform data (right column). The gray values in the point cloud subsets represent intensity and have been linearly stretched to span the available 8-bit (0-255) dynamic range. The increase in point density on VOs that can be achieved using the full-waveform data is readily apparent through comparison of the point cloud subsets in the middle and rightmost sections of the figure. This increase in point density is especially important in detection of small, low-reflectance objects, such as poles. For example, two laser points on an object may be too few to enable detection or to reliably distinguish between object and noise, while four may be sufficient (e.g. Object #24).

The next step was to investigate the ability to automatically eliminate false alarms and correctly identify required objects in Steps 3 and 4 of our workflow. This was performed by analyzing the input and output to Step 3 (OIS analysis) and comparing the final output from Step 4 (manual QA/QC and attribution) against the results obtained using the field-surveyed reference data. The input to Step 3 consisted of 531 registered detections output from Step 2, including both correct detections of real objects and false alarms (“false objects”). These detections were analyzed against the OIS, depicted in Fig. 7, using the SML software. Software variables were set to only output objects that penetrated, or came within 1.5 m of penetrating the OIS, resulting in 529 objects being automatically eliminated and 2 being flagged as penetrating. The output

objects were then attributed using the DSS imagery in ArcGIS (Step 4). The final results were compared with the output generated by running the reference data through NGS' OIS analysis software, and were found to match exactly: in both final data sets, Objects 17 and 18 were listed as penetrating and attributed as antennas. No "false objects" needed to be analyzed and eliminated in Step 4, as they were all automatically eliminated in Step 3 (OIS analysis).

The final analysis step was to compare the total time required to complete the obstruction survey, as well as the required human time, with the corresponding times from NGS' 2003 lidar obstruction survey of Stafford Regional Airport (RMN) (described in Parrish et al. 2004). (This comparison focused solely on the times required to survey obstructions and generate final obstruction lists with object-type attributes; times required to complete other aspects of airport surveys, such as profiling runways and geolocating navigational aids (FAA 1996; FAA 2008) that were outside the scope of this study were not considered.) The results of the comparison clearly demonstrated the efficiency of our workflow; the decrease in total (end-to-end) obstruction survey time was approximately 46%, and the decrease in required human time was 38%.

Discussion and Conclusions

Following a decade of research by NGS and others into the capability to perform airport obstruction surveys using lidar, we have developed and tested a new workflow with the goal of making such surveys not only feasible, but also efficient and cost-effective. One key to the workflow presented here is the use of the full-waveform data, which was shown to enable generation of dense, detailed point clouds, which are well suited for this application. Equally important is the ordering of steps in the workflow, in particular the placement of the VO

detection step before the OIS analysis and QA/QC. This sequence was shown to enable the detection threshold to be set very conservatively, minimizing the probability of a missed obstruction. Additionally, the amalgamation of steps requiring manual analysis minimizes the amount of human time in the process, and the overall workflow has been designed to reduce the total survey time. The 46% decrease in total (end-to-end) obstruction survey completion time and 38% decrease in required human time, as compared with the most recent NGS lidar obstruction survey, are quite encouraging.

Future work being planned by NGS includes conducting additional tests at two airports in different parts of the United States with differing terrain and land cover/land use types, testing and comparing different VO detection algorithms in Step 2 of the workflow described in this paper, and continuing to refine the procedures for production use. It is anticipated that these efforts, building upon the results presented here, will enable implementation of a wide-scale, production lidar airport obstruction surveying program in the very near future.

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Table 1: Mission parameters for Madison, WI, airborne lidar data acquisition.

| Mission Parameter | Value |
|----------------------------------|--|
| Dates of data acquisition | June 23-24, 2006 |
| Aircraft | Dynamic Aviation Beechcraft King Air |
| Flying speed over ground | 70 m/s (135 kts) |
| Survey altitude | 800 m (AGL) |
| Lidar system | Optech ALTM 3100 w/ waveform digitizer |
| Pulse energy | 80 - 90 μ J @ 70 kHz |
| Pulse width | 11 ns (+/- 2 ns) |
| Laser rep rate | 70 kHz |
| Beam divergence | Narrow (0.3 mrad) |
| Laser wavelength | 1064 nm |
| Scan angle | +/- 17.3° |
| Scan rate | 49.8 Hz |
| Swath width | 500 m |
| Nominal horizontal point spacing | 0.7m x 0.7m |

Table 2: Mission parameters for Madison, WI, digital imagery acquisition.

| Mission Parameter | Value |
|------------------------------|-------------------------------------|
| Dates of data acquisition | September 14-15, 2006 |
| Aircraft | NOAA Cessna Citation |
| Flying speed over ground | 77 m/s (150 kts) |
| Survey altitude | 1,370 m (AGL) |
| Camera | Applanix DSS 322 |
| Focal length | 60 mm |
| Pixel size | 9 μ m |
| CCD array dimensions | 4092 x 5436 pixels (along x across) |
| Ground Sample Distance (GSD) | 21 cm |
| Bands | RGB |
| Endlap | 60% |
| Sidelap | 30% |
| Footprint | 840 x 1120 m (along x across) |

Table 3: Comparison of number of points on objects in point clouds generated from discrete-return lidar data and full-waveform lidar data.

| ID | Object Type | UTM Easting (NAD 83) | UTM Northing (NAD 83) | # Points on Object in Discrete-Return Lidar Point Cloud | # Points on Object in Full-Waveform Lidar Point Cloud |
|----------------|---------------------|-----------------------------|------------------------------|--|--|
| 01 | Cell tower | 298238.92 | 4769245.46 | 20 | 29 |
| 02 | Tree coniferous | 298323.75 | 4769282.46 | 136 | 541 |
| 03 | Tree deciduous | 297218.01 | 4769221.56 | 317 | 1235 |
| 04 | Tree deciduous | 298376.03 | 4768978.59 | 549 | 2351 |
| 05 | Tree deciduous | 297154.63 | 4769264.68 | 951 | 4990 |
| 06 | Tree deciduous | 297416.02 | 4769158.35 | 315 | 1374 |
| 07 | Light pole | 297503.01 | 4769288.81 | 8 | 41 |
| 08 | Tree deciduous | 297332.61 | 4769177.18 | 98 | 400 |
| 09 | Cell tower | 296317.79 | 4769552.05 | 8 | 34 |
| 10 | Light pole | 297456.10 | 4769321.14 | 9 | 14 |
| 11 | Tree deciduous | 296213.83 | 4769644.49 | 293 | 1437 |
| 12 | Antenna | 296288.65 | 4769550.51 | 5 | 9 |
| 13 | Tree deciduous | 295423.81 | 4769253.43 | 153 | 389 |
| 14 | Tree deciduous | 295333.60 | 4769357.72 | 829 | 3860 |
| 15 | Transmission tower | 295060.06 | 4770004.02 | 12 | 24 |
| 16 | Tree deciduous | 295130.66 | 4769980.78 | 86 | 418 |
| 17 | Antenna | 293549.21 | 4770122.78 | 1738 | 5564 |
| 18 | Antenna | 297455.24 | 4769449.05 | 1199 | 3383 |
| 19 | Antenna | 298189.71 | 4769599.46 | 56 | 214 |
| 20 | Microwave antenna | 298160.65 | 4769634.38 | 158 | 608 |
| 21 | Antenna | 298279.73 | 4769606.85 | 321 | 1669 |
| 22 | Antenna on tower | 296916.12 | 4769233.56 | 61 | 107 |
| 23 | Antenna on building | 296991.22 | 4769252.49 | 39 | 81 |
| 24 | Light pole | 296073.54 | 4769882.00 | 2 | 4 |
| 25 | Transmission tower | 295626.76 | 4769814.53 | 34 | 47 |
| 26 | Cell tower | 294237.25 | 4770413.94 | 65 | 72 |
| 27 | Steeple | 294596.31 | 4769571.97 | 107 | 331 |
| 28 | Water tank | 294877.82 | 4768931.97 | 728 | 2492 |
| 29 | Light pole | 296369.84 | 4770666.86 | 11 | 44 |
| 30 | Tree deciduous | 297787.87 | 4770050.27 | 312 | 1241 |
| 31 | Building | 298191.35 | 4769865.84 | 4827 | 13164 |
| 32 | Tree deciduous | 296428.16 | 4770535.24 | 64 | 343 |
| 33 | Tree deciduous | 296527.90 | 4770263.49 | 153 | 552 |
| 34 | Light pole | 298322.94 | 4770609.67 | 5 | 8 |
| 35 | Building | 298361.09 | 4770728.58 | 868 | 4016 |
| Average | | | | 415 | 1460 |

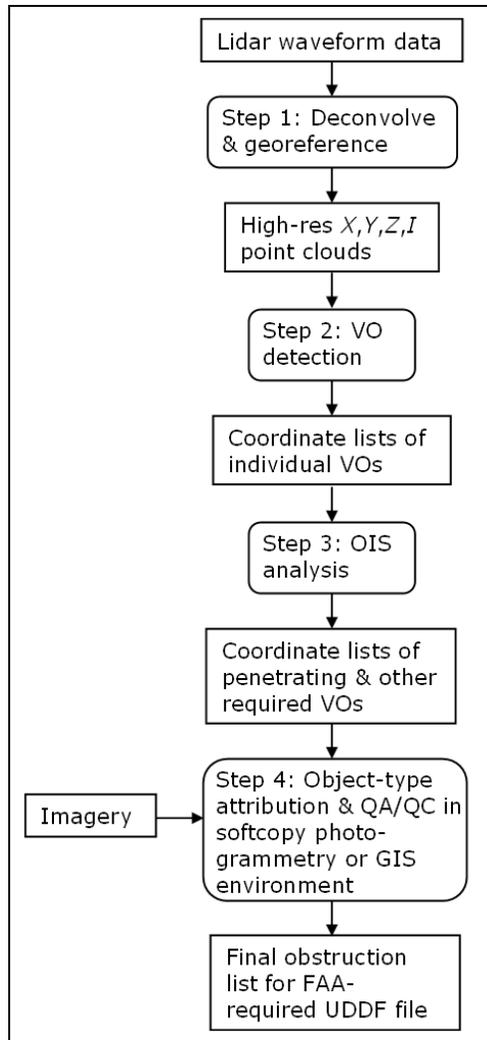


Fig. 1. Workflow diagram for lidar airport obstruction surveying using full-waveform data.

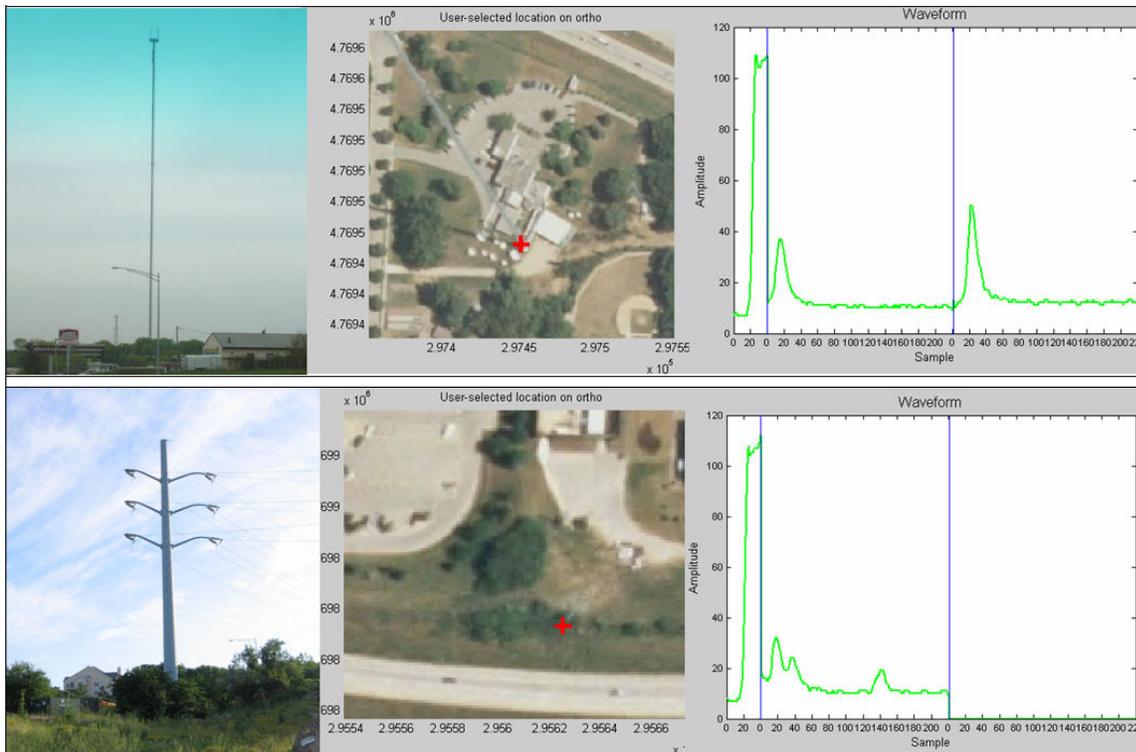


Fig. 2. Examples of digitized lidar waveforms.

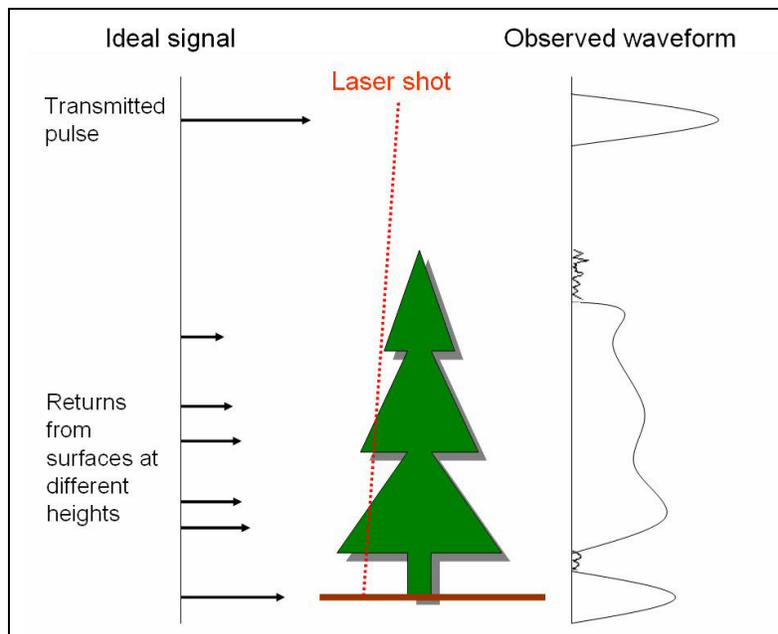


Fig. 3: Illustration of an “ideal” lidar waveform consisting of a train of spikes in time (left) and a more realistic representation of an actual observed waveform (right).

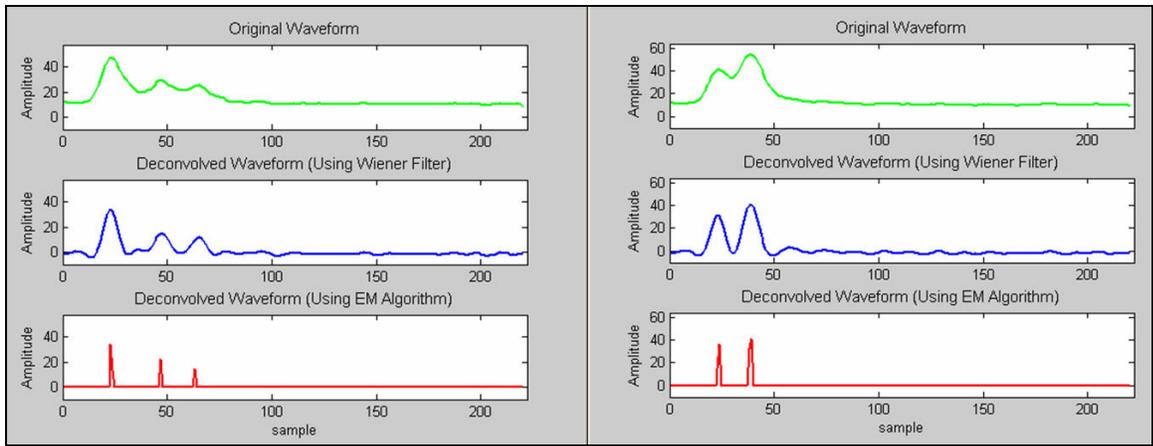


Fig. 4. Examples of two observed lidar waveforms (top, green), Wiener-filtered versions (middle, blue), and output of EM deconvolution algorithm (bottom, red).

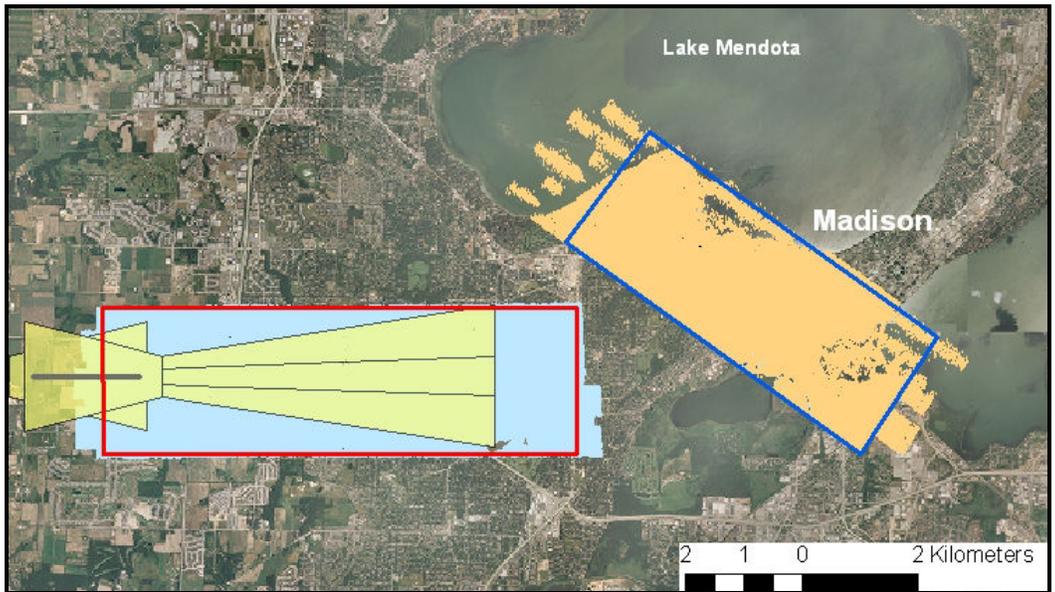


Fig. 5. Madison, Wisconsin project areas. The full-waveform lidar coverage is shown in orange and light blue. The simulated ANA LPV approach for the southwest project site is shown in yellow.

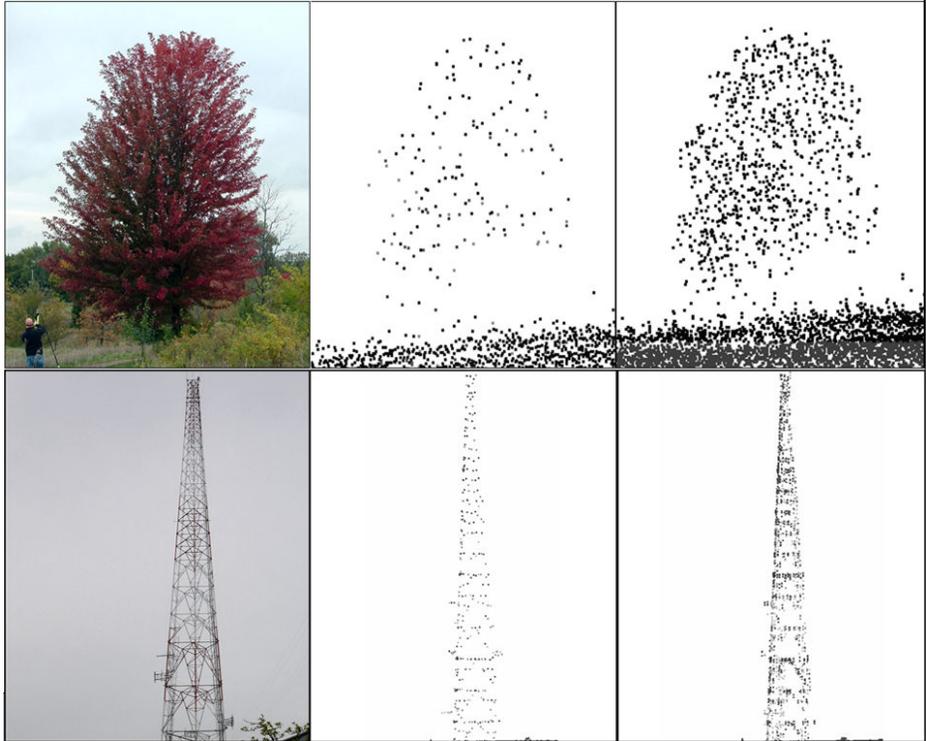


Fig. 6. Comparison of point density for two different VOs, a tree (top) and tower (bottom), in point clouds generated using the discrete-return lidar data (middle) and full-waveform lidar data (right). Photos of the two objects are shown at left. The gray values in the point cloud subsets represent intensity, and have been linearly stretched to take advantage of the available 8-bit dynamic range. The achievable increase in point density using full-waveform data is clearly seen through this comparison.



Fig. 7. OIS analysis for simulated runway and ANA LPV approach in southwest Madison project area. The red points indicate locations of field-surveyed reference objects.